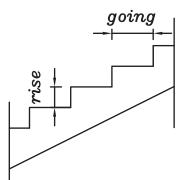
STAIRS

-هى بلاطات مائلة للتوصيل من منسوب الى منسوب اخر وتكون مزودة بدرج لامكانية الصعود عليها وعادة تكون بلاطة السلم اما (one way or cantilever solid slab)



-وتتراوح ابعاد(rise) من (mse) وعادة تؤخذ (se) وتتراوح ابعاد (going) من (mse) وتتراوح ابعاد (30cm)

- أكبر عدد من الدرجات فاللبة الواحدة يساوى (١٤) درجة

No. of steps= difference between levels

Rise height (0.15)

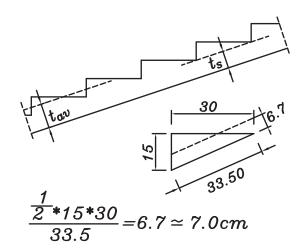
Length of stair=No. of steps *0.30

Concrete dimensions:

$$t_s = \frac{L_s}{25.30.36}$$
 (One way slab)

$$t_s = \frac{L_c}{10}$$
 (cantilever slab)

$$t_{av} = t_s + 7 cm$$

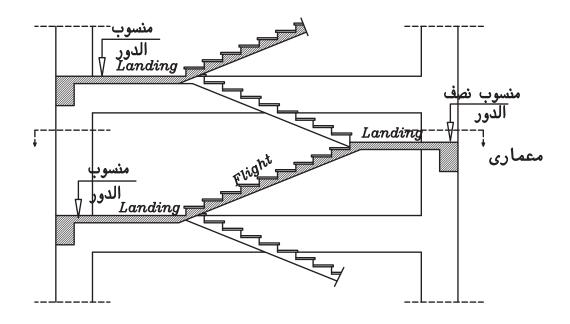


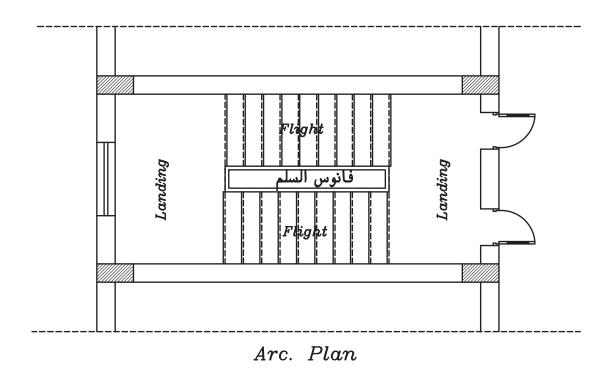
وعادة تؤخد (30cm) ·

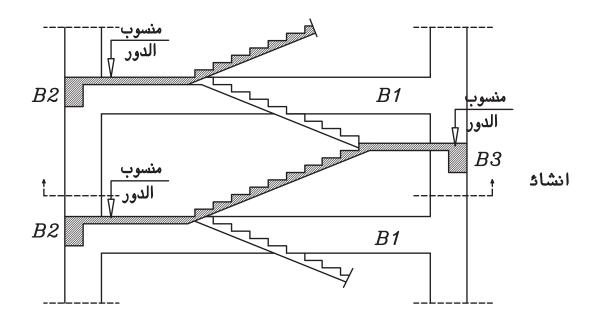
 $w_{su}=1.4(t_{av} v_c + F.C.) + 1.6L.L.\cos\theta kN/m^2$

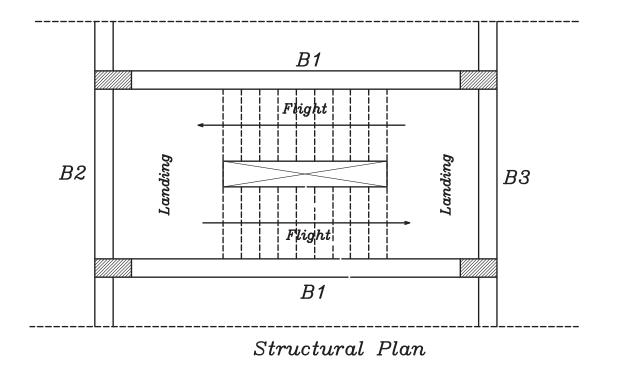
ـ ملحوظة

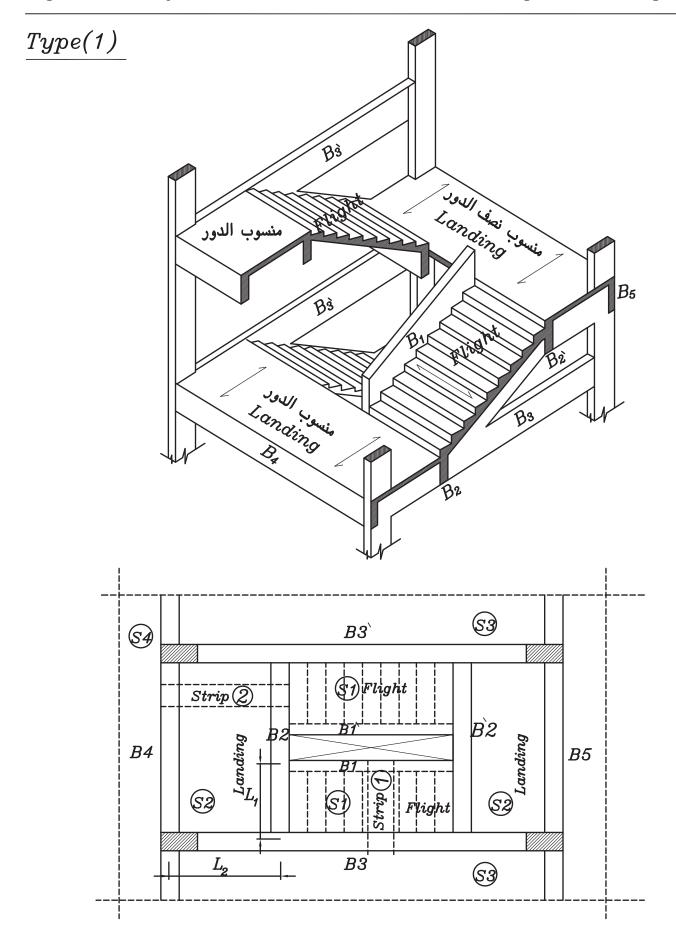
لاحظ ان $(t_{\rm s})$ هى التى تستخدم فى التصميم بينما $(t_{\rm s})$ تستخدم فى حساب الاحمال فقط \cdot











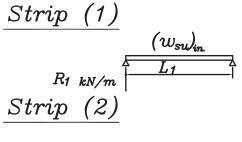
1-Slabs

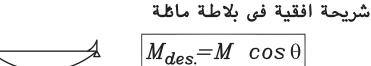
$$Calculate \ t_s$$

$$t_{av} = t_s + 7cm$$

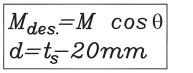
$$w_{su} = 1.4[t_s \gamma_c + F.C.] + 1.6L.L.$$

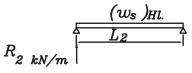
$$w_{\!s\!u}\!\!=\!$$
 1.4 $[t_{\!a\!v}\gamma_c\!\!+\!F.C.]\!\!+\!$ 1.6 $L.L.\cos\theta$







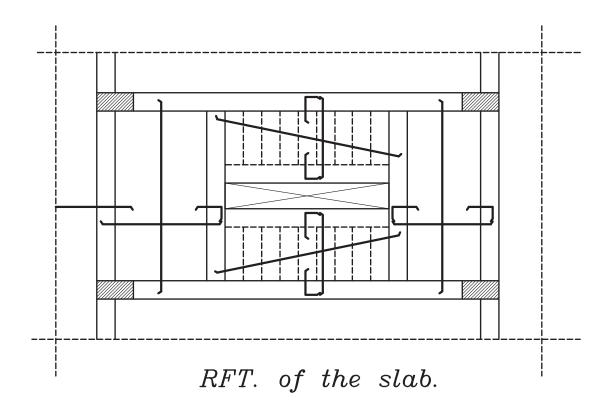




$$M$$
 $B.M.D.$

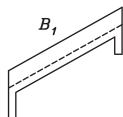
$$M_{des.} = M$$
 $d = t_s - 20mm$

منحوطة نصمم شريحة السلم على انها simple span ودلك لسهولة الحل والتنفيد



2–Beams

B_1 , B_1



$$w_1 = 0.w. + R_1$$

 $w_1 = \gamma_c b(t - t_s) * 1.40 + R_1$

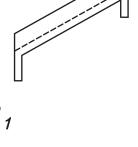
$$B_2$$
 , B_2

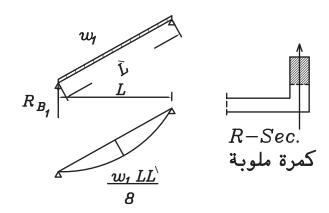
$$w_2 = o.w. + R_2$$

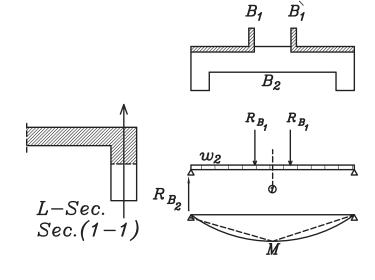
$$w_2 = \gamma_c b(t - t_s) * 1.40 + R_2$$

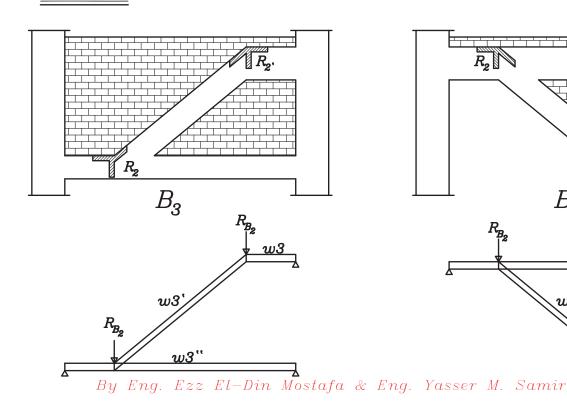
$\underline{L-Sec}$.

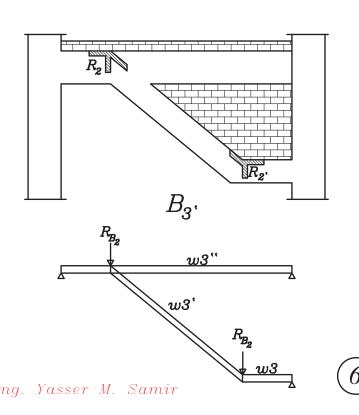
$$egin{aligned} C.L. &- C.L. \ B &= 6\,t_{S}+b \ &K\,rac{L}{10}+b \end{aligned}$$
 الاقل







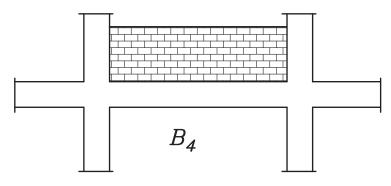




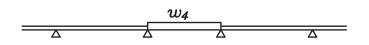
$$w_3 = o.w. + walls$$

 $w_3 = o.w. + walls + R_1$
 $w_3 = o.w. + walls + S_3$

$\underline{\underline{B_4}}$ $w_4 = 0. w + walls + R_2 + S_4$

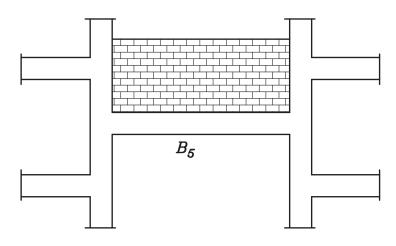


Continous Beam.

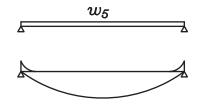


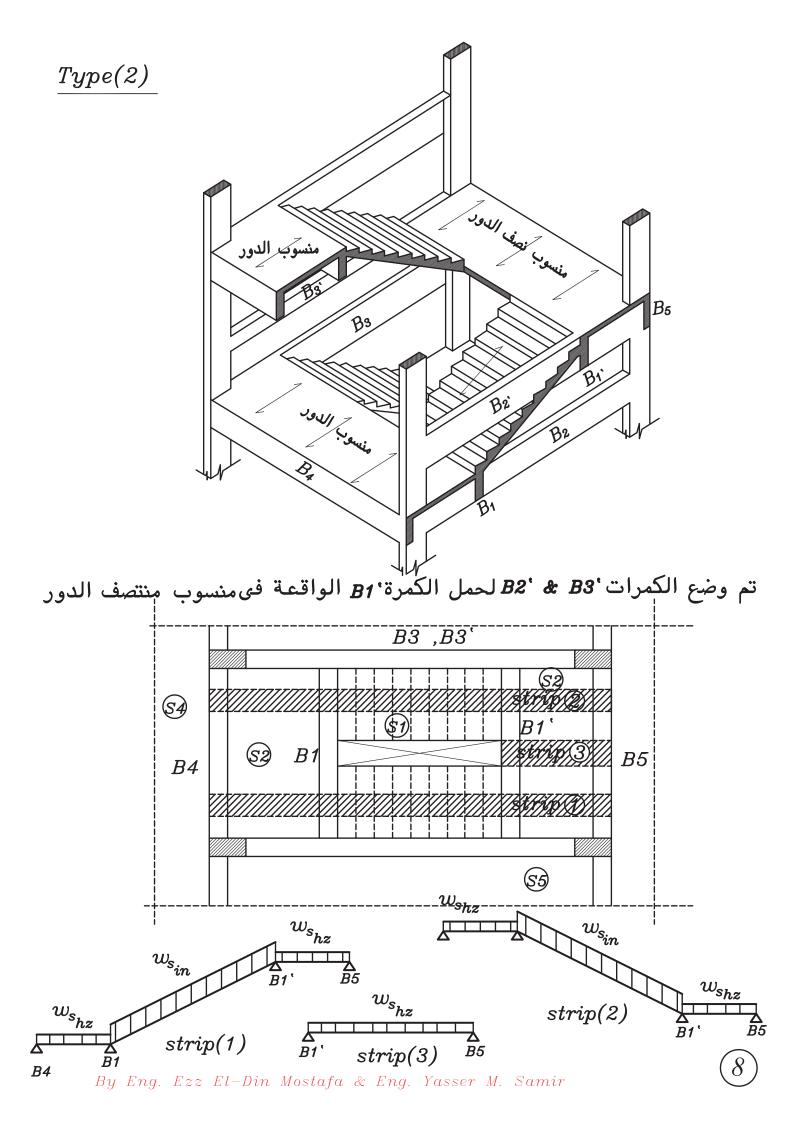
B_5

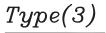
 $w_5 = o.w. + walls + R_2$

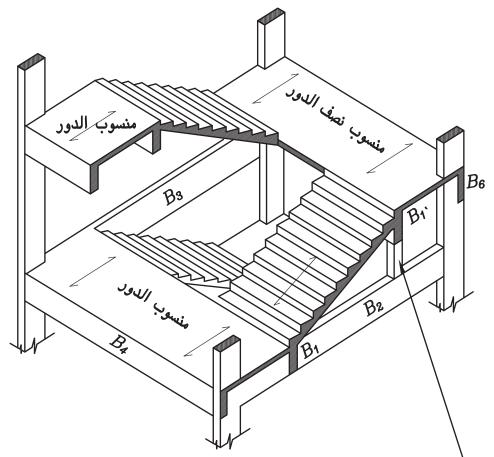


 B_5 Simple Beam.

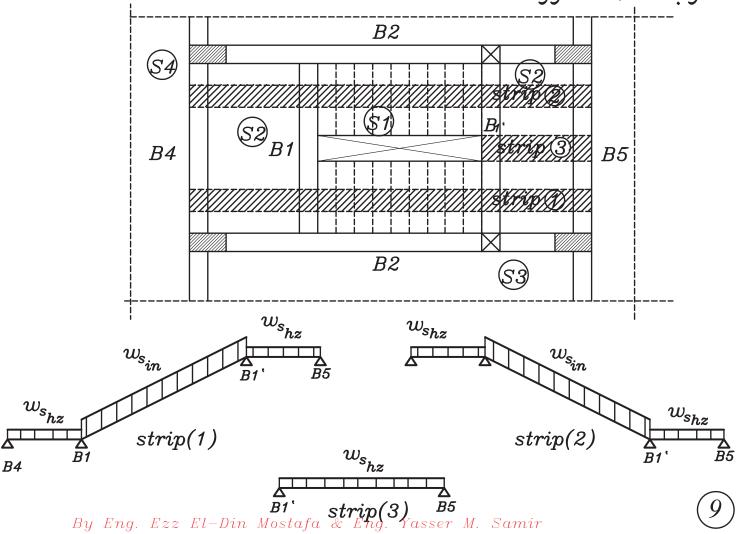


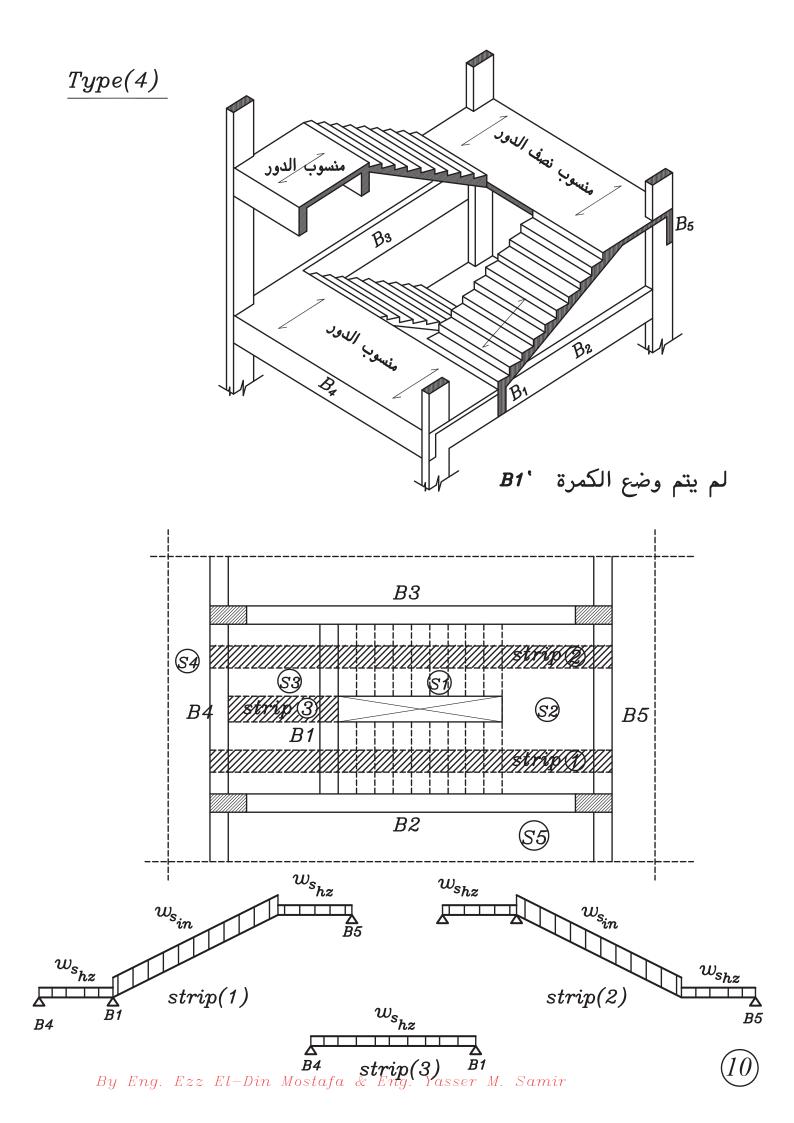


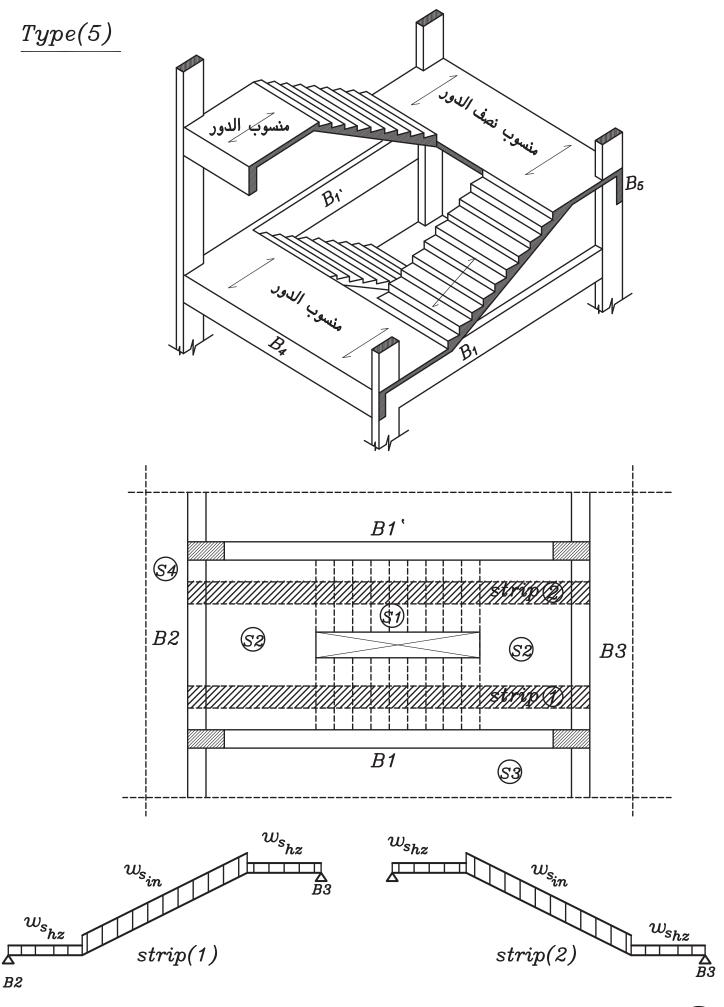


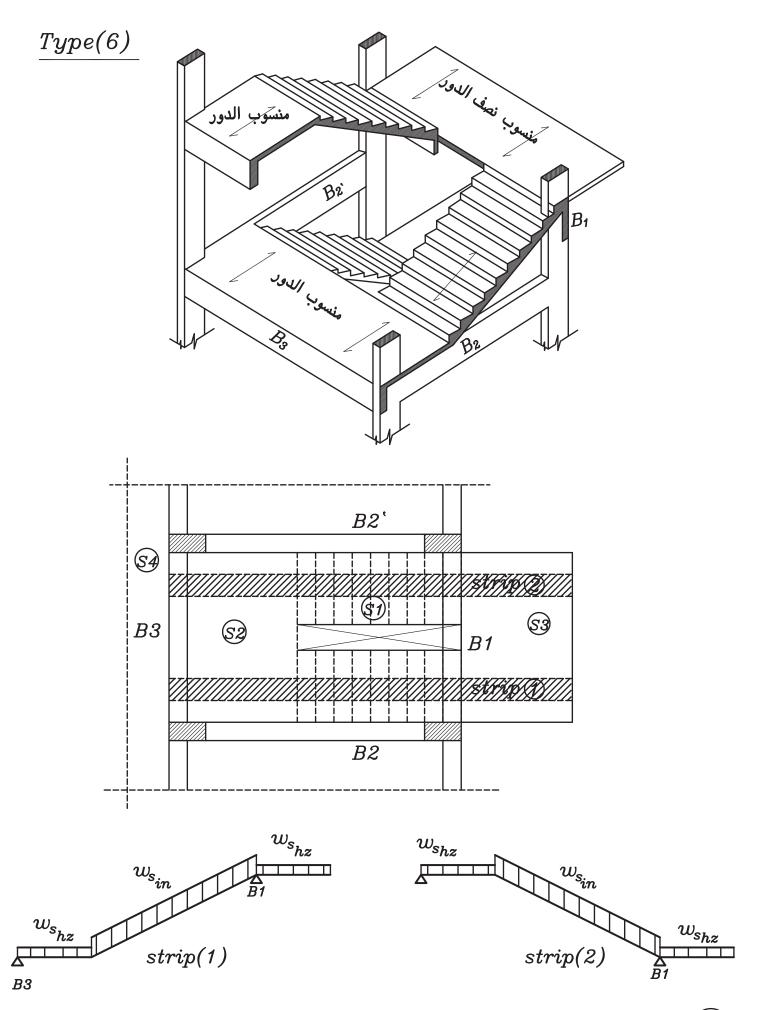


تم وضع P_{ost} بدلا من الكمرات '83 B وذلك لحمل الكمرة 'B الواقعة في منسوب منتصف الدور

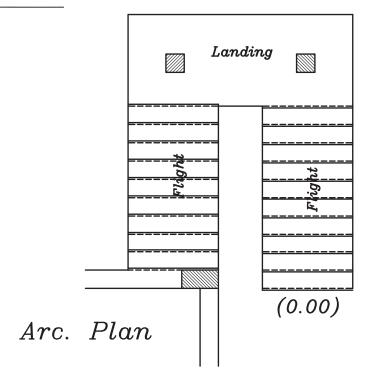


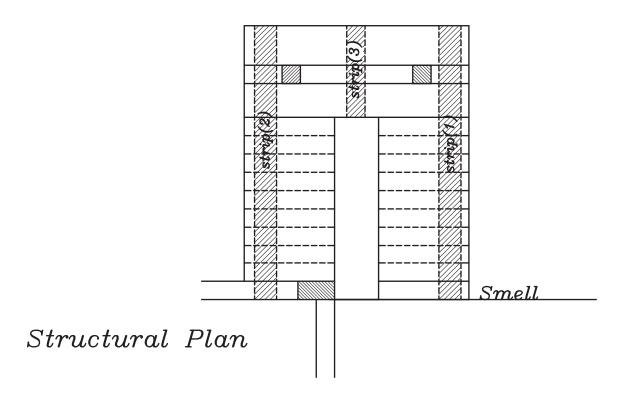


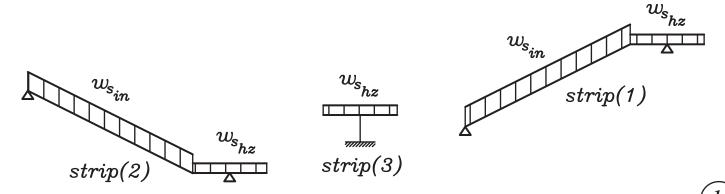


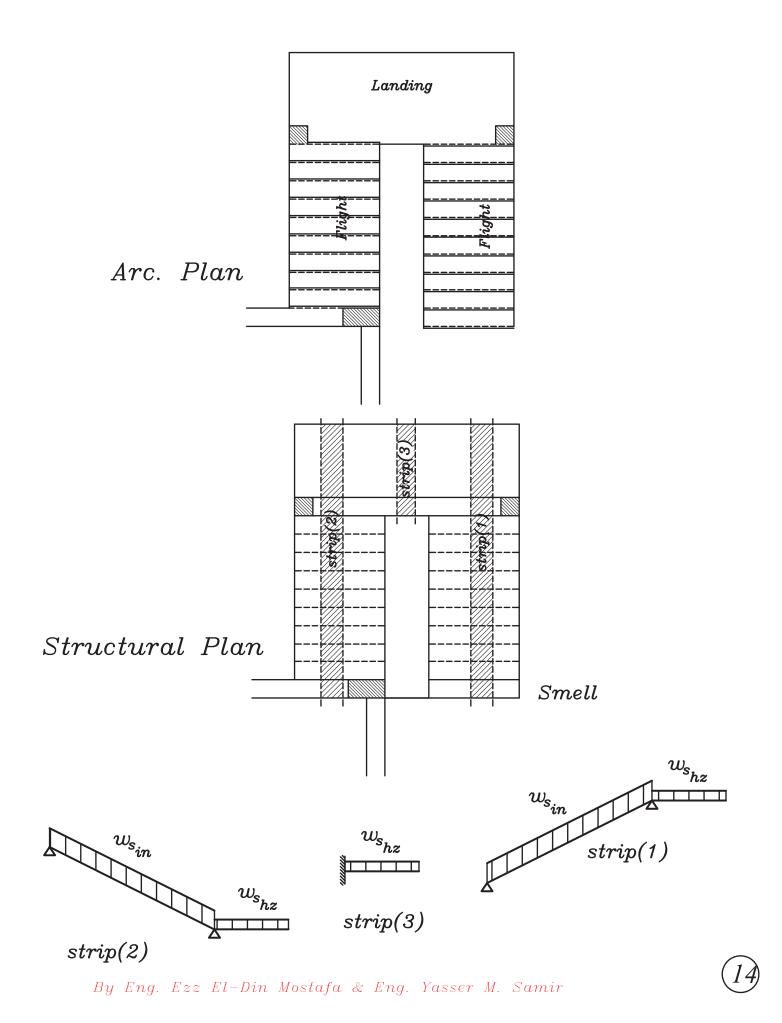


Example 1:- Show How to solve this stair

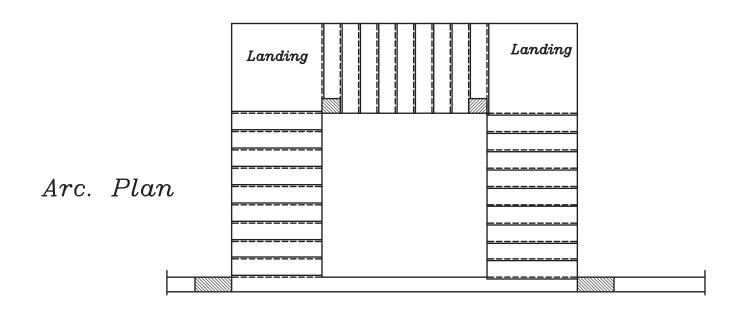


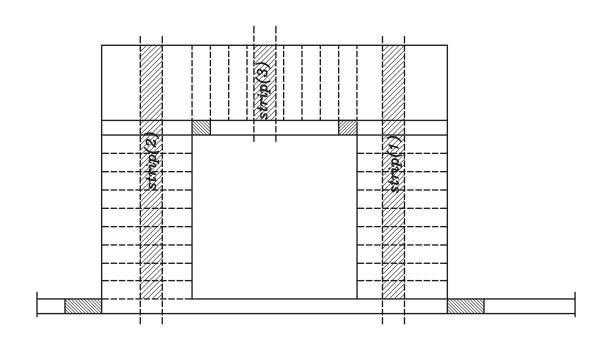


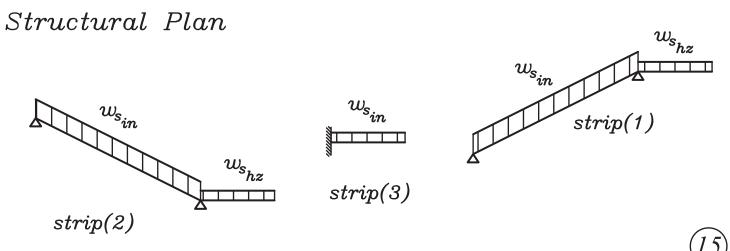




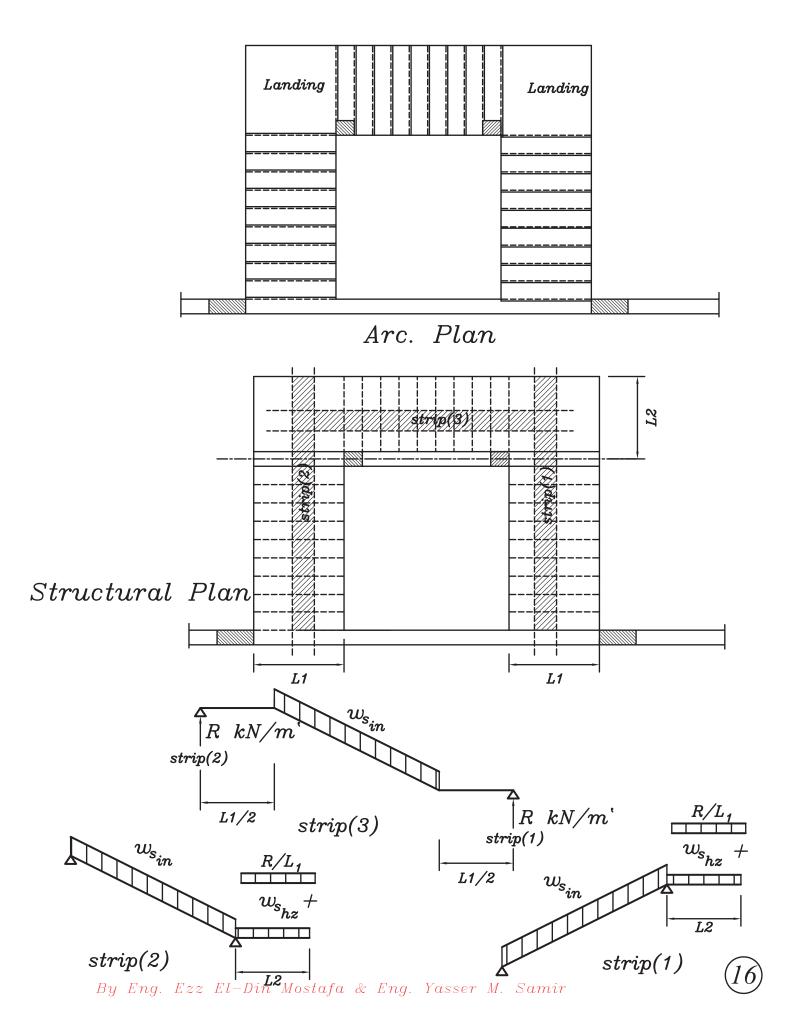
Example 3:- Show How to solve this stair



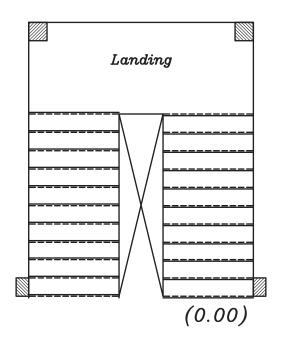




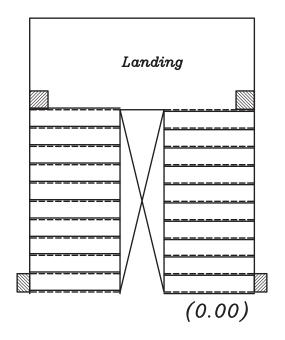
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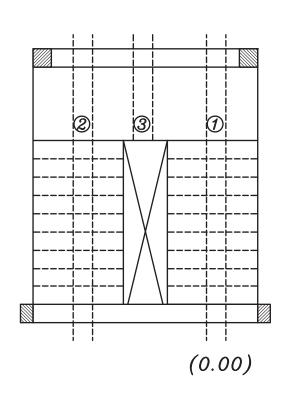
Example 4:- Show How to solve this stairs



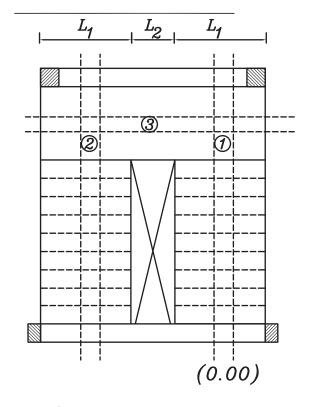
Case 1:-

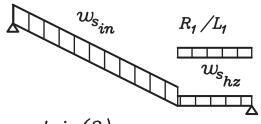


Case 2:-

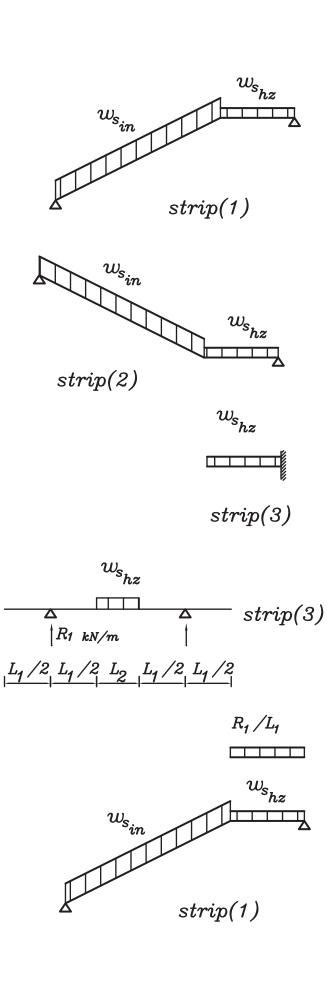


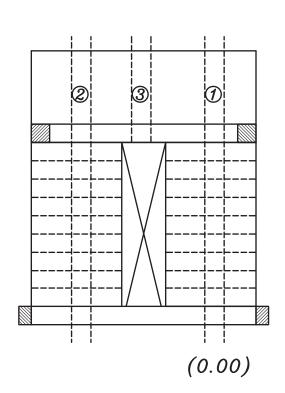
Alternative solution



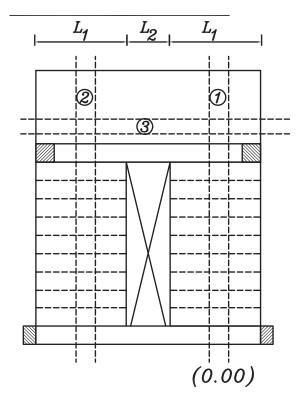


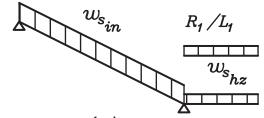
strip(2)

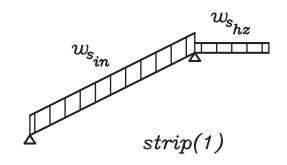


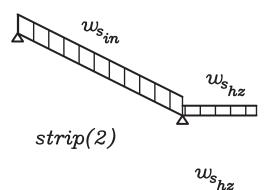


Alternative solution



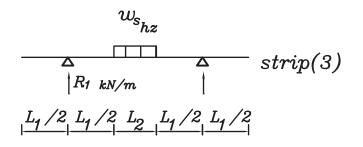


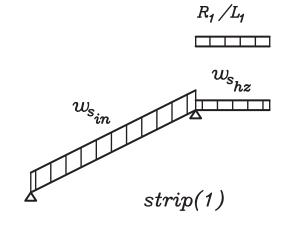




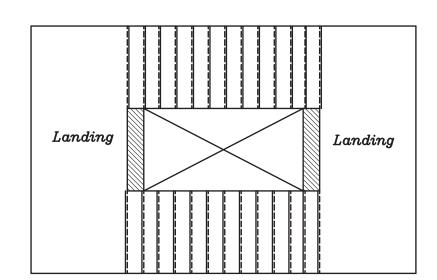


strip(3)

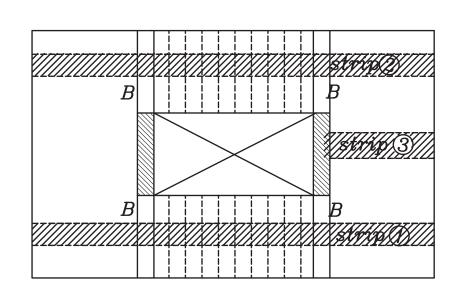




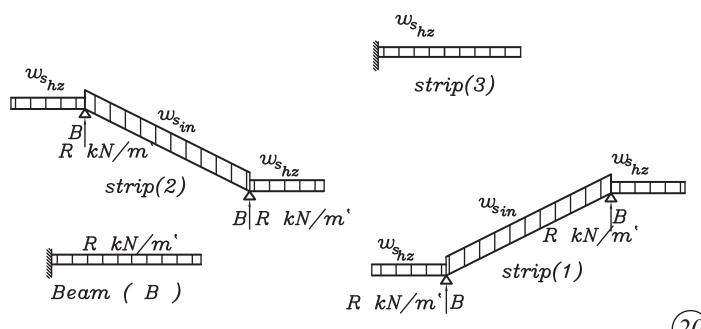
Example 5:- Show How to solve this stair



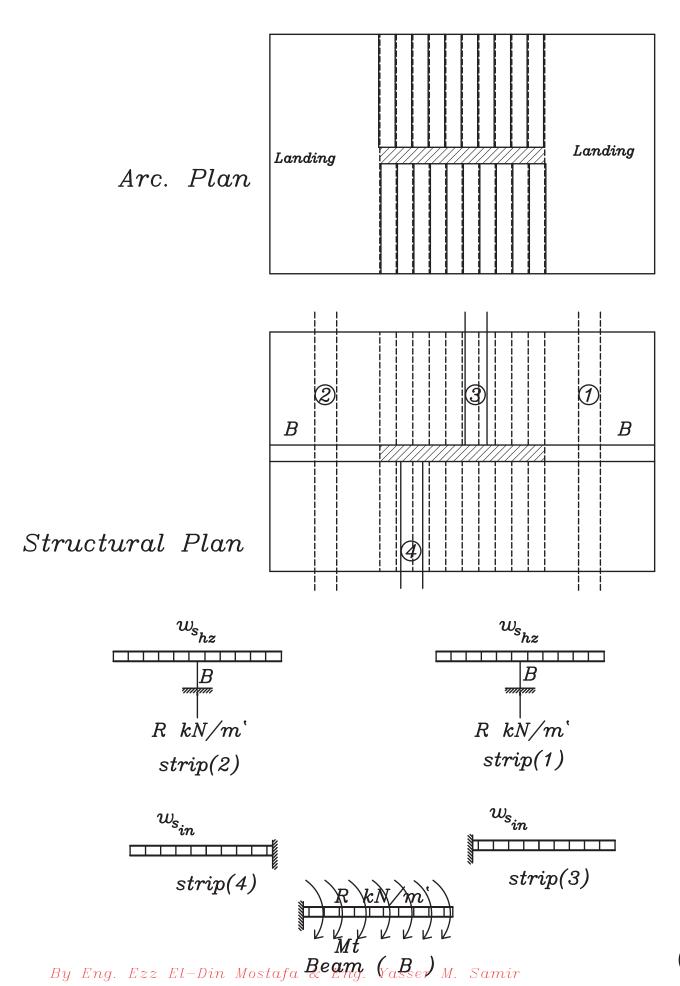
Arc. Plan

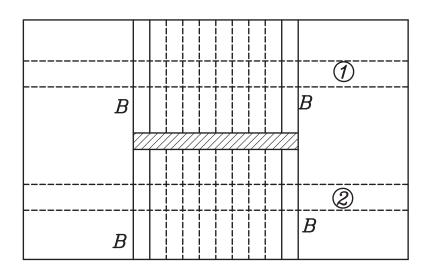


Structural Plan

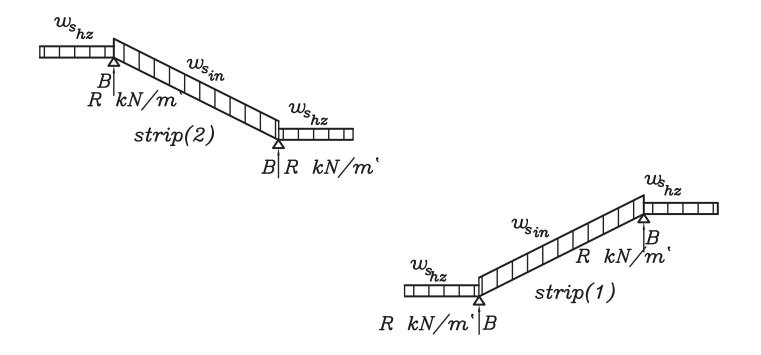


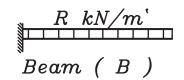
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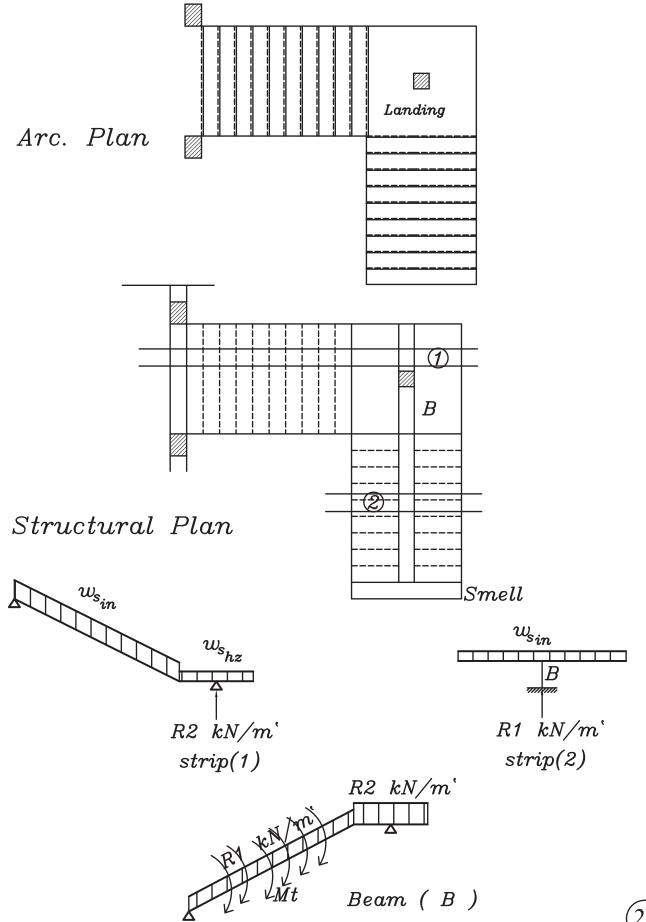




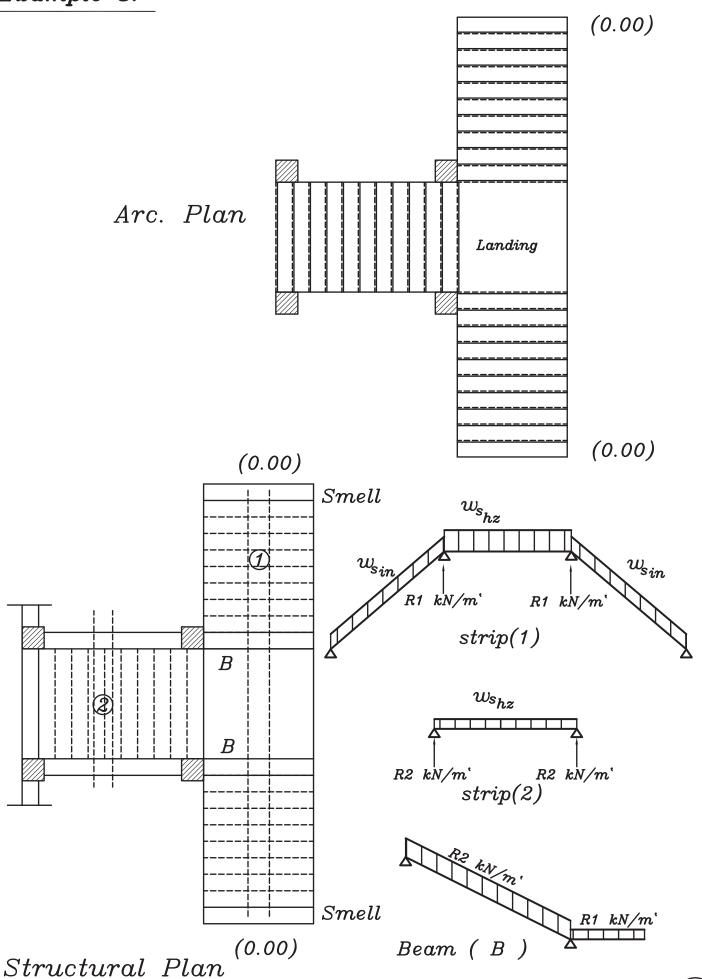
Structural Plan



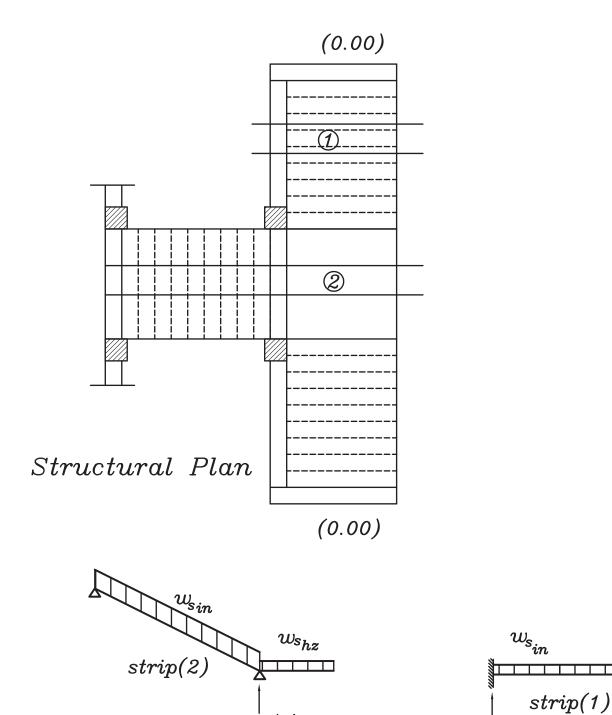


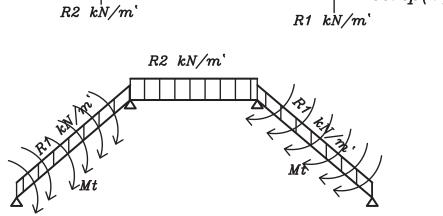


Example 8:- Show How to solve this stair



Another solution

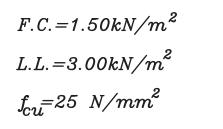




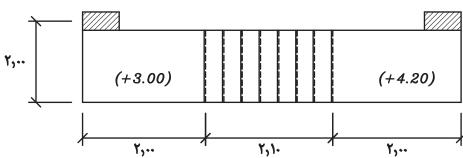
Example 9:-

The following figure show the general layout of stair cases (each step 300x150), proceed with the following:

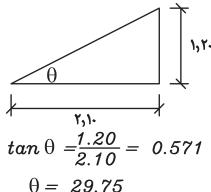
- 1- Complete design including all slabs and their supporting beams
- 2- Draw to a convenient scale the details of reinforcement of the stair and the supporting beams in plan & sections.



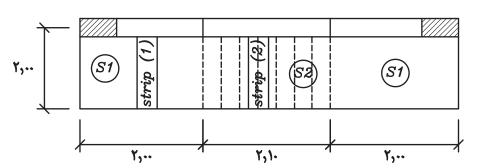
Steel used 360/520



0.w of beam=4.0kN/m







1-slab thickness

$$t_{\rm s} = \frac{L_{\rm c}}{15} + 20 = \frac{2000}{15} + 20 = 153.33 \, \text{mm}$$

$$t_s = \frac{L_c}{15} + 20 = \frac{2000}{15} + 20 = 153.33mm$$

$$\Longrightarrow$$
 Take $t_s = \underline{16}$ cm (check deflection)

$$t_{av} = t_s + 7 = 23 \text{ cm}$$

2-Calculation of load

$For \ Landing :-$

$$w_{shz} = 1.4[t_s \gamma_c + F.c.] + 1.6L.L.$$
 kN/m^2

$$w_{shz} = 1.4[0.16*25+1.50]+1.6*3.00$$

$$w_{\rm shz} = 12.50 \ kN \backslash m^2$$

$For \ Flight :-$

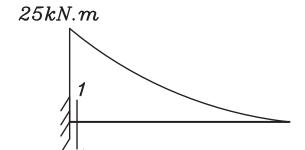
$$w_{sin} = 1.4[t_{sav} \gamma_c + F.c.] + 1.6L.Lcos\theta kN/m^2$$

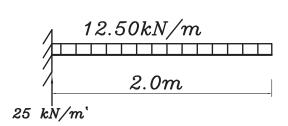
$$w_{sin} = 1.4[0.23*25+1.50]+1.6*3.00*cos(29.75)$$

$$w_{\sin} = 14.32 \ kN \backslash m^2$$

-Design of sections

Strip (1) :-





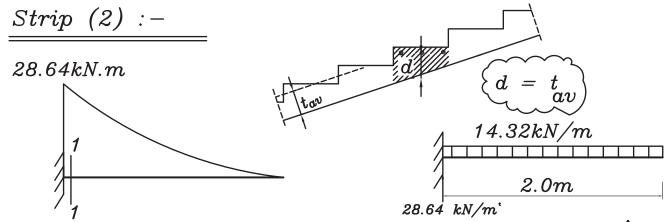
Sec(1-1)

$$M_{u.l.} = 25.00kN.m$$
 & $B = 1000mm$ & $d = 160 - 20 = 140mm$

$$140 = C_1 \sqrt{\frac{25.00*10^6}{1000*25}} \qquad C_1 = 4.43 \quad \& \quad J = 0.813$$

$$A_s = \frac{25.00*10^6}{0.813*140*360} = 610.13 \text{ mm}^2/\text{m}^3$$

$$A_s = 5 \, \phi \, 13/m'$$



ـ ملحوظة

لاحظ أننا لا نعتبر هذة الشريحة شريحة أفقية في بلاطة مائلة لأن العزم سالب

$$\underline{Sec(1-1)} \qquad \underbrace{d = t \atop av}$$

وعمودى على أتجاة الحديد لذلك فمو لا يحتاج الى تحليل

$$M_{u.l.} = 28.64kN.m$$
 &

$$B = 1000 mm$$

$$B = 1000mm$$
 & $d = 230$ mm

$$230 = C_1 \sqrt{\frac{28.64*10^6}{1000*25}}$$
 $C_1 = 6.8 \& J = 0.826$

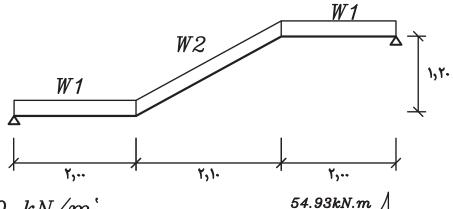
$$C_1 = 6.8 \& J = 0.826$$

$$A_{s} = \frac{28.64*10^{6}}{0.826*230*360} = 418.76 \text{ mm}^{2}/\text{m}'$$

$$A_s / step = 418.76*0.3 = 125.63 \text{ mm}^2 / step.$$

 $A_s = 2 \phi 10/\text{step}$

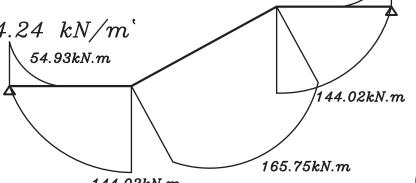
-Design of Beam



$$W1 = 4*1.4+25=30.60 \text{ kN/m}$$

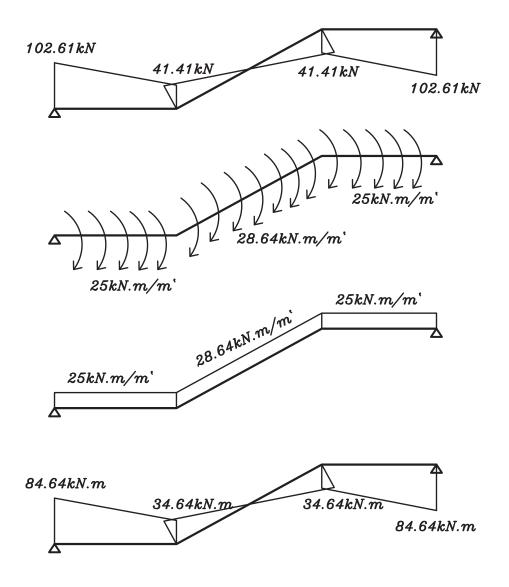
W2 = 4*1.4+28.64=34.24 kN/m





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سہبی



 $Take\ b = 400\ mm$

1-from bending
$$\implies d=3.5\sqrt{\frac{165.75*10^6}{400*25}}=451.00$$
 mm
$$d=500 \text{ mm} \qquad t = d+50\text{mm} = \underline{550 \text{ mm}}$$

$$2-from \text{ Torsion } \implies \frac{3M_{tu}}{b^2 t} = 2 \text{ N/mm}^2 \implies \frac{3*84.64*10^6}{400^2* t} = 2.00$$

 $t = 793.5 \ mm \implies Take \ t = 800 \ mm$

 $Take_{By} \ \underset{Eng.\ Ezz}{\textit{Eeam}} = (400\ x\ 800\)_{Yasser\ M.\ Samir}$

Design for shear+Torsion:

$$q_{su} = \frac{Q_u}{bd} = \frac{102.61*10}{400*750}^3$$
 $q_{su} = 0.342 \ N/mm^2$

$$q_{tu} = \frac{M_{tu}}{1.7A_{oh} t_{e}}$$

$$A_{oh} = (400-50)(800-50)$$

$$A_{oh} = 262500mm^{2}$$

$$P_{h} = 2[350+750]=2200mm$$

$$t_{e} = \frac{A_{oh}}{P_{h}} = 119.32mm$$

$$q_{tu} = \frac{84.64*10^{6}}{1.7*262500*119.32}$$

$$q_{tu} = 1.59 N/mm^{2}$$

$$q_{cu} = 0.24\sqrt{25/1.5} = 0.98N/mm^2$$

$$q_{u_{max}}^{=0.70\sqrt{25/1.5}} = 2.86N/mm^2$$

$$\sqrt{q_{su}^2 + q_{tu}^2} = \sqrt{(0.34)^2 + (1.59)^2} = 1.63 \ N/mm^2 < q_{u_{max}}$$
 (ok)

(Design for Torsion)

$$A_{str} = \frac{M_{tu}.S}{1.7A_{oh}(\frac{f_{y_{st}}}{\gamma_s})}$$

$$\implies \frac{A_{str}}{S} = \frac{84.64*10^6}{1.7*262500*360/1.15}$$

$$\frac{A_{str}}{S} = 0.606$$

assume $A_{str} = \#10 = 78.5 mm^2$

$$\frac{78.5}{S} = 0.606 \implies S=129.54 \text{ mm} > 100 \text{ mm}$$

No. of
$$stirrups/m' = \frac{1000}{S} = 7.72$$

Take stirrups 8 \$\psi\$10/m' (2 branches)

$$A_{sL} = (\frac{A_{str}}{S})*(P_h)*(\frac{f_{y_{st}}}{f_{y_{l,h}}})$$

$$A_{sL} = 0.606*2200*\frac{360}{360} = 1333.2mm^{2}$$

$$\frac{A_{sL}}{4} = \frac{1333.2}{4} = 333.3 \text{ mm}^2$$

Design for B.M.

Sec(1-1) (L-sec)

$$B = \begin{bmatrix} 6*0.16+0.4=1.36 & m \\ C.L. \rightarrow C.L. = 2.0+0.2=2.20m \\ \frac{1*6.42}{10} +0.40=1.04m \end{bmatrix}$$

 $B=1040mm \& d=800-50mm=750mm \& M_u=165.75 kN.m$

$$750 = C_1 \sqrt{\frac{165.75*10^6}{1040*25}} \quad C_1 = 9.4 \quad \& \quad J = 0.826$$

$$A_s = \frac{165.75*10^6}{0.826*750*360} = 743.21 \text{mm}^2$$

$$A_{s} = \frac{165.75*10^{6}}{0.826*750*360} = 743.21 \, \text{mm}^{2}$$

$$A_{s_{min}} = \frac{1.1}{f_y} bd = \frac{1.1}{360} 400*750 = 916.67 mm^2 > A_s$$

$$\frac{1.1}{f_y} bd = \frac{1.1}{360} 400*750=916.67mm^2$$

$$A_{s_{min}} = 1.3A_{s_{reg}} = 1.3*743.21=966.173mm^2$$

$$\frac{0.15}{100} bd = \frac{0.15}{100} 400*750=450.00mm^2$$

$$A_{s_{min}} = 916.67 mm^2$$

$$A_{s_{total}} = A_{s_{(B.M.)}} + \frac{A_{sL}}{4}$$

= 916.67+333.3=1250.0 mm²

$$A_{s_{total}} = 5 \oplus 18$$

Sec(2-2) (R-sec)

 $b=400mm \& d=800-50mm=750mm \& M_u=54.93 \ kN.m$

$$750 = C_1 \sqrt{\frac{54.93*10^6}{400*25}} \quad C_1 = 10.12 \quad \& \quad J = 0.826$$

$$A_s = \frac{54.93*10^6}{0.826*750*360} = 246.30 \text{mm}^2$$

$$A_{s_{min}} = \frac{1.1}{f_y} bd = \frac{1.1}{360} 400*750 = 916.67 mm^2 > A_s$$

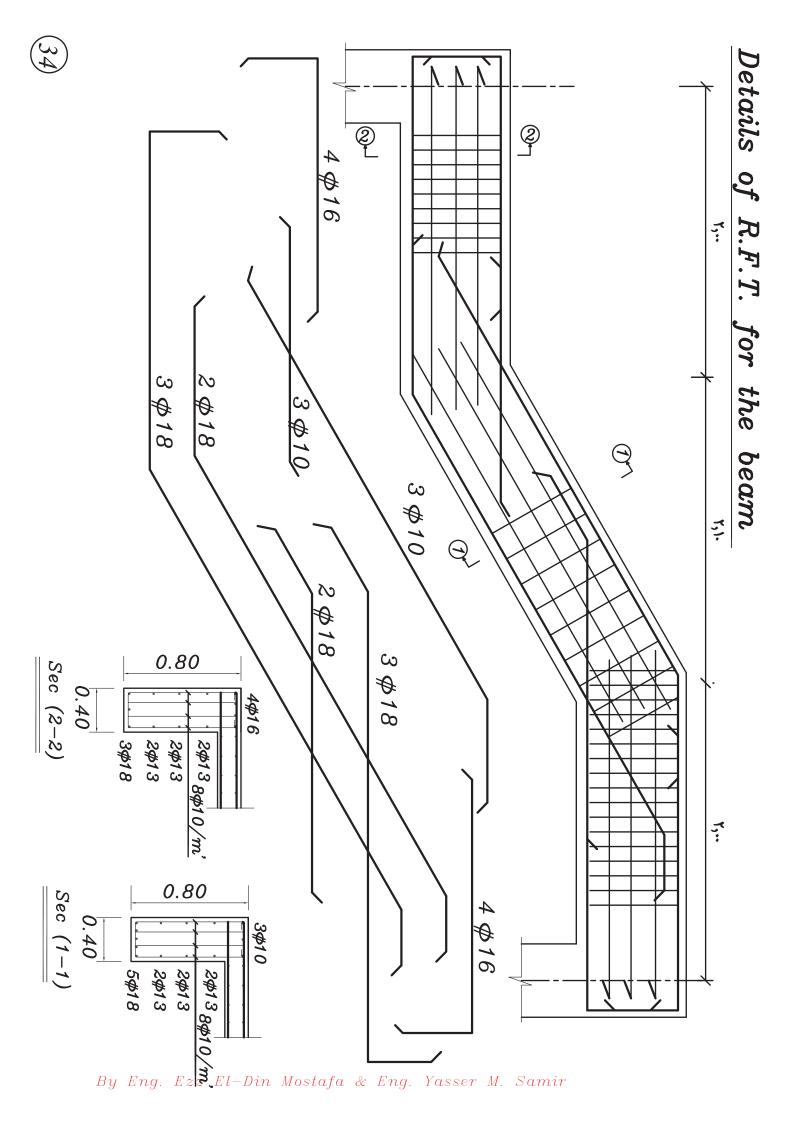
$$A_{s_{min}} = \begin{bmatrix} \frac{1.1}{f_y} & bd & = \frac{1.1}{360} & 400*750 = 916.67mm^2 \\ 1.3A_{s_{req}} = 1.3*246.30 = 320.20 & mm^2 \\ \frac{0.15}{100} & bd & = \frac{0.15}{100} & 400*750 = 450.00mm^2 \end{bmatrix}$$

$$A_{s_{min}} = 450.00 mm^2$$

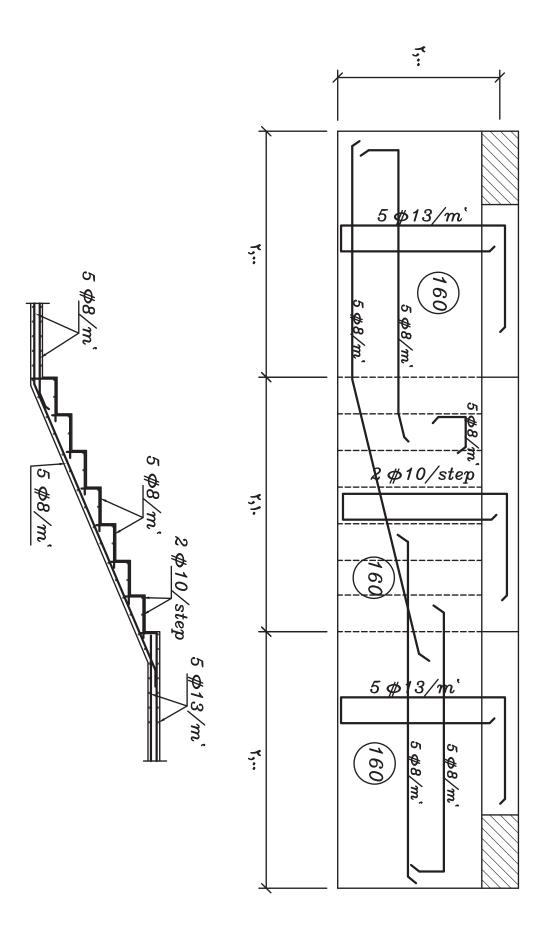
$$A_{s_{total}} = A_{s_{(B.M.)}} + \frac{A_{sL}}{4}$$

= $450.00 + 333.3 = 783.30 \quad mm^{2}$

$$A_{s_{total}} = 4 \oplus 16$$







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Example 10:-

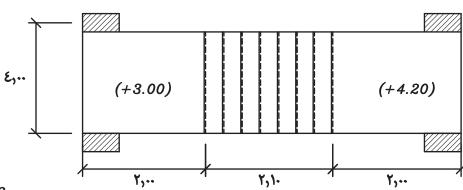
The following figure show the general layout of stair cases (each step 300x150), proceed with the following :

- 1 Complete design including all slabs and their supporting beams
- 2- Draw to a convenient scale the details of reinforcement of the stair and the supporting beams in plan & sections.

$$F.C.=1.50kN/m^{2}$$

 $L.L.=3.00kN/m^{2}$
 $f_{cu}=25 \ N/mm^{2}$

Steel used 360/520 0.w of beam=4.0kN/m

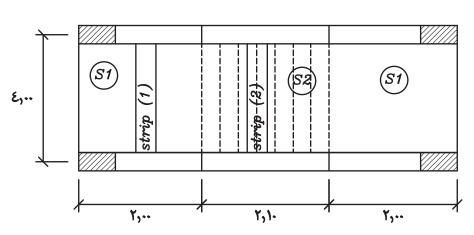


 $\tan \theta = \frac{1.20}{2.10} = 0.571$ $\theta = 29.75$



for one way slab S1

Solution



$$t_{\rm s} = \frac{L}{25} = \frac{400}{25} = 16cm$$

$$t_s = \frac{L}{25} = \frac{400}{25} = 16cm$$

$$\Longrightarrow$$
 Take $t_s = 16$ cm

$$t_{av} = t_s + 7 = 23$$
 cm

2-Calculation of load

$For \ Landing :-$

$$w_{shz} = 1.4[t_s \gamma_c + F.c.] + 1.6L.L.$$
 kN/m^2

$$w_{shz} = 1.4[0.16*25+1.50]+1.6*3.00$$

$$w_{\rm shz} = 12.50 \ kN \backslash m^2$$

For Flight :-

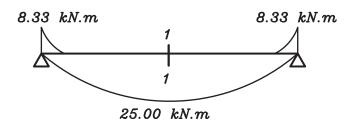
$$w_{sin} = 1.4 [t_{sav} \gamma_c + F.c.] + 1.6 L.L \cos\theta kN/m^2$$

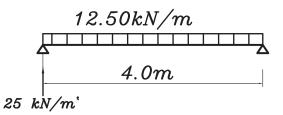
$$w_{sin} = 1.4[0.23*25+1.50]+1.6*3.00*cos(29.75)$$

$$w_{sin} = 14.32 \ kN \backslash m^2$$

-Design of sections

Strip (1) :-





Sec(1-1)

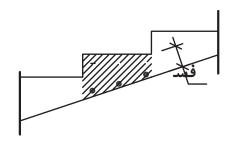
$$M_{u.l.} = 25.00kN.m$$
 & $B = 1000mm$ & $d = 160 - 20 = 140mm$

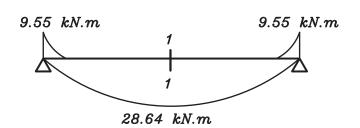
$$140 = C_1 \sqrt{\frac{25.00*10^6}{1000*25}}$$
 $C_1 = 4.43$ & $J = 0.82$

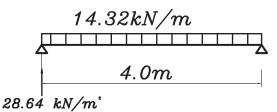
$$A_s = \frac{25.00*10^6}{0.82*140*360} = 607.58 \text{ mm}^2/\text{m}^2$$

$$A_s = 6 \phi 12/m'$$

Strip (2) :-

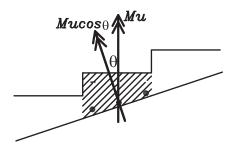






. ملحوظة

لاحظ أننا هنا نعتبر هذة الشريحة شريحة أفقية فى بلاطة مائلة لأن العزم موجب وماثل على أتجاة الحديد لذلك فهو يحتاج الى تحليل



$$Sec(1-1)$$

$$M_{u.l.} = 28.64 kN.m$$
 & $B = 1000 mm$ & $d = 160 - 20 = 140 mm$

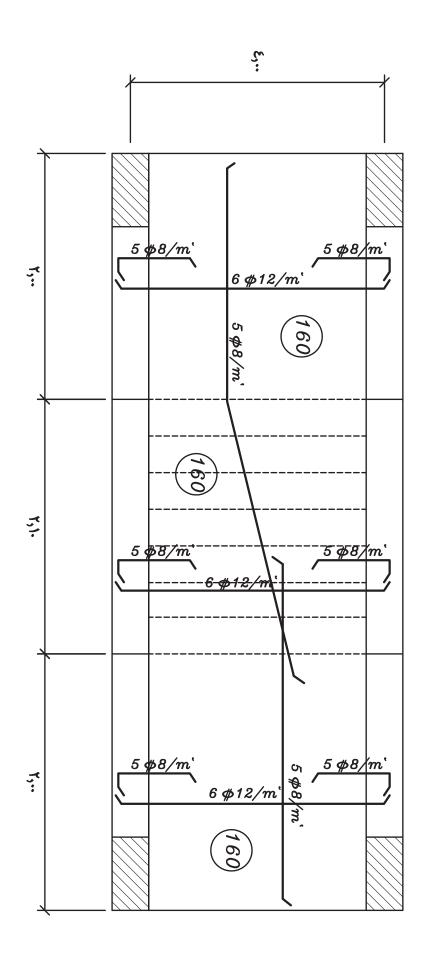
$$Mdes=28.64cos \theta = 24.87 \ kN.m$$

$$140 = C_1 \sqrt{\frac{24.87*10^6}{1000*25}} \qquad C_1 = 4.44 \quad \& \quad J = 0.82$$

$$A_s = \frac{24.87*10^6}{0.82*140*360} = 604.20 \quad mm^2/m'$$

$$A_s = 6 \not \oplus 12/m'$$





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